

**Amendments to the Claims:**

The following listing of claims will replace all prior versions, and listings, of claims in the application:

1. (Currently Amended) A combustor for a gas turbine ~~engine comprising~~  
engine, comprising:

a combustion chamber wall having formed therein at least one hole for  
admitting air into the combustion chamber;

at least one air intake chute aligned with said hole;

during operation a hoop stress field having regions of high and low stress  
concentration around said hole;

wherein said chute is ~~attached~~ provided with a flange disposed around one end  
thereof, and at least one tab that projects from the outer edge of said flange and is welded to  
the combustor wall in a region of low stress-concentration. concentration along the outer edge  
of the at least one tab such that the at least one tab provides a guide for the location and  
length of the weld.

2. (Original) A combustor as claimed in claim 1 wherein the chute is attached to  
the combustor wall in at least two regions of low stress concentration.

3. (Original) A combustor as claimed in claim 2 wherein areas where the chute  
is attached to the combustor wall are substantially in the same radial plane.

4. (Original) A combustor as claimed in claim 2 wherein the areas of attachment  
are provided on diametrically opposite sides of said chute.

5. (Canceled)

6. (Original) A combustor as claimed in claim 5 wherein the flange is circular.

7-8. (Canceled)

9. (Original) A combustor as claimed in claim 7 wherein the at least one tab projects from the edge of the flange up to about  $0.14 \times$  (flange diameter).
10. (Original) A combustor as claimed in claim 7 wherein the at least one tab has a length of up to about  $0.25 \times$  (flange diameter) of the diameter of the flange.
- 11-15. (Canceled)